Abstract #1678



English

Leveraging Terrestrial Industry for Utilization of Space Resources

NASA's Journey to Mars: Pioneering Next Steps in Space Exploration released in October of 2015 states that NASA is working toward the capability to work, operate, and sustainably live safely beyond Earth. To progress from our current "Earth-Reliant" approach to exploration and eventually become "Earth Independent", we need to first identify resources in space and then learn to use and harvest them to minimize logistics from Earth, reduce costs, and enable sustainable and affordable space transportation and surface operations. Known as In Situ Resource Utilization (ISRU), the collection and conversion of space resources into products such as propellants, fuel cell reactants, and life support consumables can greatly reduce the mass, cost, and risk of space exploration. Also, the ability to perform civil engineering, construction, and manufacturing at sites of exploration can also allow for increased crew safety and sustainable growth in critical infrastructure. Much of what NASA wants to do on the Moon and Mars with respect to harnessing and utilizing space resources has been performed and perfected on Earth over the centuries. While minimizing mass and operating in the vacuum of space may be unique challenges to NASA, both terrestrial industry and NASA face many of the same challenges associated with operating in severe environments, minimizing maintenance and logistics, maximizing performance per unit mass and volume, performing remote and autonomous operations, and integrating hardware from many vendors and countries. In the end, both NASA and terrestrial industry need to obtain a return on the investment for the development and deployment of these capabilities. This paper will first examine what is ISRU and what are the space resources of interest. The paper will than discuss what are NASA's approach, life cycle, and economic considerations for implementing ISRU. The paper will outline the site and infrastructure needs associated with a phased implementation of ISRU into human missions to the Moon and Mars. T

French

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Mr. Stanley Starr (UnknownTitle) NASA Kennedy Space Center Gerald Sanders has worked at the NASA Johnson Space Center (JSC) full time for almost 30 years, primarily in the Propulsion and Power Division of the Engineering Directorate, and has extensive experience in chemical propulsion, fluid systems, and In Situ Resource Utilization (ISRU). Mr. Sanders currently serves as the ISRU Capability Leadership Team lead for the Agency in the Office of Chief Engineer, and Deputy Manager for the ISRU Technology project. Mr Sanders currently serves as a Co-Investigator on the Mars OXygen Isru Experiment (MOXIE) for the Mars 2020 rover, and initiated development of the RESOLVE payload for the Resource Prospector mission scheduled for launch for 2021. He has also served as the ISRU lead or deputy on multiple human architecture team studies and international coordination groups. Mr. Sanders has worked in the area of In-Situ Resource Utilization (ISRU) for over 21 years. Besides working ISRU, Mr. Sanders has worked extensively in the area of crewed vehicle and in-space chemical propulsion development, and served as the Chief and Deputy Chief for the Propulsion and Fluid Systems Branch. Mr. Sanders received a Bachelor's of Science degree in Aerospace Engineering from the University of Cincinnati in June of 1987.

CIM | TPMS |

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Leveraging Terrestrial Industry for Utilization of Space Resources

Gerald B. Sanders, NASA/JSC, Diane L. Linne, NASA/GRC Stanley O. Star, NASA/KSC Dale Bougher, Deltion Innovatio Presentation to CIM/Planetary and Terrestrial Mining Sciences Symposium May, 2017

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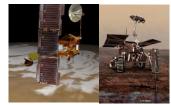
- What is *In Situ* Resource Utilization (ISRU) and what are the space resources of interest?
- What are the approach, life cycle, and economic considerations for implementing ISRU?
- What are the site and infrastructure needs and implementation phasing for ISRU?
- What are the terrestrial industries and operations that are synergistic with ISRU?
- What are the challenges and similarities between ISRU and Terrestrial Industry that can be exploited?
- Where do we go from here?





ISRU involves any hardware or operation that harnesses and utilizes 'in-situ' resources to create products and services for robotic and human exploration

Resource Assessment (Prospecting)



Assessment and mapping of physical, mineral, chemical, and water resources, terrain, geology, and environment

In Situ Manufacturing



Production of replacement parts, complex products, machines, and integrated systems from feedstock derived from one or more processed resources

Resource Acquisition



Atmosphere collection, drilling, excavation, transfer and preparation/ beneficiation before processing

In Situ Construction



Civil engineering, infrastructure emplacement and structure construction using materials produced from *in situ* resources

➢ Radiation shields, landing pads, roads, berms, habitats, etc.

Resource Processing/ Consumable Production



Extraction and processing of resources into products with immediate use or as feedstock for construction & manufacturing

Propellants, life support gases, fuel cell reactants, etc.

In Situ Energy



Generation and storage of electrical, thermal, and chemical energy with *in situ* derived materials

➢ Solar arrays, thermal storage and energy, chemical batteries, etc.

- 'ISRU' is a capability involving multiple elements to achieve final products (mobility, product storage and delivery, power, crew and/or robotic maintenance, etc.)
- 'ISRU' does not exist on its own. By definition it must connect and tie to users/customers of ISRU products and services



Mission Variables for Implementation of ISRU



Location



Resource Location Factors



Environmental Factors



Resource Demand and Availability



Resource Extraction Method



Resource Pre-Processing and Transportation



0

Resource Processing



Time

Moon, Mars, Mars Moons, Near-Earth Asteroids

Slopes, craters, rock size/distribution, geographic location (poles, equator)

Climate (temperature, wind, season), pressure/vacuum, sunlight, gravity

Atmosphere/Gases (carbon dioxide), Water/Ice, Volatiles (hydrogen, helium), Metals (iron, nickel, titanium), Non-Metals (silicon, oxygen)

Gas separation/compression, surface regolith/ granular material mining, rock/quarry mining, subsurface mining/drilling extraction

Sorting, crushing/sizing, beneficiation; buckets, augers, conveyors, pneumatic

Electrical, chemical, thermal

Human consumables, propellants, stored energy, construction & manufacturing feedstock

Time available for product production; Time available for setup; Day/night cycle



Main *Natural* Space Resources of Interest for Human Exploration





(PSR)

Icy Regolith in

Shadowed Regions

Solar wind hydrogen

Permanently

with Oxygen

Water (Hydrogen)



Minerals in Lunar Regolith: Ilmenite, Pyroxene, Olivine, Anorthite

CO, CO₂, and HC's

Solar Wind from

Sun (~50 ppm)

(Gases)



Metals



Minerals in Lunar Regolith

in PSR

- Iron/Ti: Ilmenite
- Silicon: Pyroxene, Olivine, Anorthite
- Magnesium: Mg-rich Silicates
- Al: Anorthitic Plagioclase



Hydrated Soils/Minerals: Gypsum, Jarosite, Phylosilicates, Polyhdrated Sulfates

Subsurface Icy Soils in Mid-latitudes to Poles

Carbon Dioxide in the atmosphere (~96%)

Carbon Dioxide in the atmosphere (~96%)

Minerals in Mars Soils/Rocks

- Iron: Ilmenite, Hematite, Magnetite, Jarosite, Smectite
- Silicon: Silica, Phyllosilicates
- Aluminum: Laterites, Aluminosilicates, Plagioclase
- Magnesium: Mg-sulfates, Carbonates, & Smectites, Mg-rich Olivine



Subsurface Regolith on C-type Carbonaceous Chondrites

Minerals in Regolith on S-type Ordinary and Enstatite Chondrites

Hydrocarbons and Tars (PAHs) in Regolith on C-type Carbonaceous Chondrites

Minerals in Regolith/Rocks on S-type Stony Iron and M-type Metal Asteroids

Uses

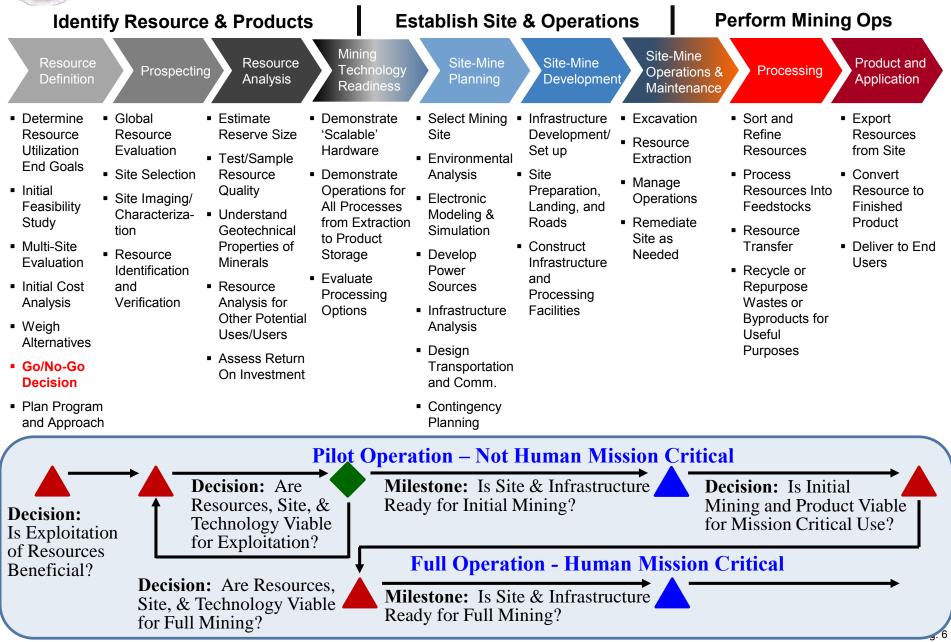
- Drinking, radiation shielding, plant growth, cleaning & washing
- Making Oxygen and Hydrogen
- Breathing
- Oxidizer for Propulsion and Power
- Fuel Production for Propulsion and Power
- Plastic and Petrochemical Production
- In situ fabrication of parts
- Electical power generation and transmission

Note: Rare Earth Elements (REE) and Platinum Group Metals (PGM) are not driving Resources of interest for Human Exploration



ISRU Implementation Life Cycle









A 'Useful' Resource Depends on the <u>Location</u>, <u>What is needed</u>, <u>How much is needed</u>, <u>How often it is needed</u>, and <u>How difficult is it to extract the resource</u>

Location

- Resource must be assessable: slopes, rock distributions, surface characteristics, etc.
- Resource must be within reasonable distance of mining infrastructure: power, logistics, maintenance, processing, storage, etc.

Resource extraction must be 'Economical'

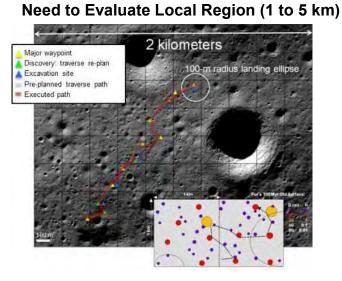
- Concentration and distribution of resource and infrastructure needed to extract and process the resource must allow for Return on Investment (ROI) for:
 - Mass ROI mass of equipment and unique infrastructure compared to bringing product and support equipment from Earth. Impacts number and size of launch vehicles from Earth
 - 1 kg delivered to the Moon or Mars surface = 7.5 to 11 kg launched into Low Earth Orbit
 - Cost ROI cost of development and certification of equipment and unique infrastructure compared to elimination of launch costs or reuse of assets (ex. reusable vs single use landers)
 - Time ROI time required to notice impact of using resource: extra exploration or science hardware, extended operations, newly enabled capabilities, etc.
 - Mission/Crew Safety ROI increased safety of product compared to limitations of delivering product from Earth: launch mass limits, time gap between need and delivery, etc.
- Amount of product needed must justify investment in extraction and processing
 - Requires long-term view of exploration and commercialization strategy to maximize benefits
 - Metric: mass/year product vs mass of Infrastructure
 - Transportation of product to 'Market' (location of use) must be considered
 - Use of product at extraction location most economical



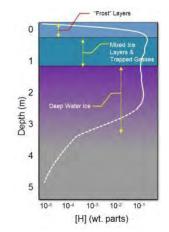
Economics of ISRU for Space Applications (2)



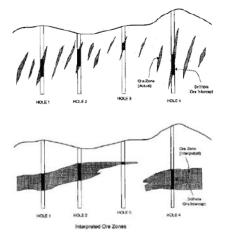
Need to assess the extent of the resource 'ore body'



Need to Determine Vertical Profile



Need to Determine Distribution



Need to assess What is needed, How much is needed, How often it is needed

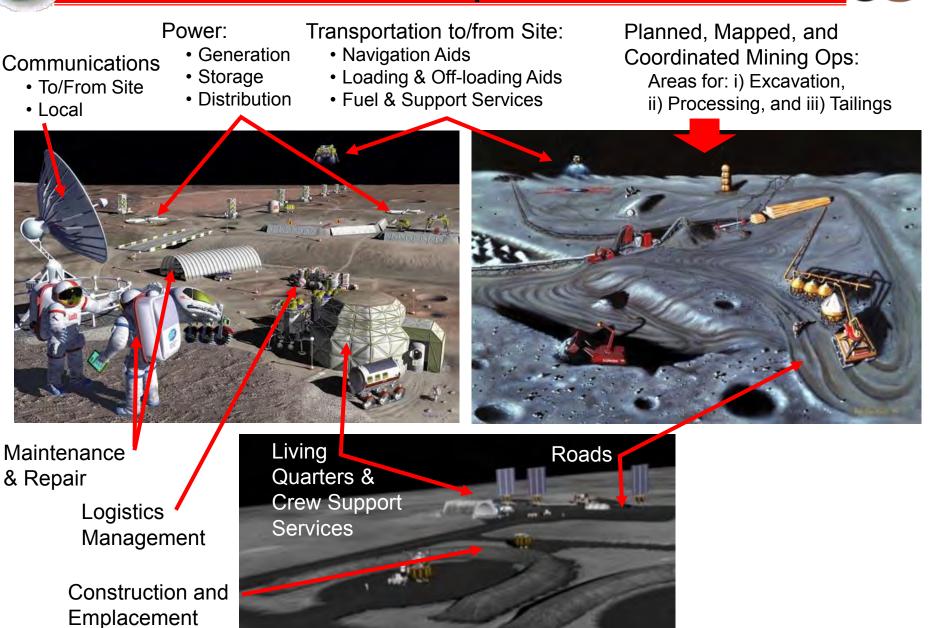
Resource Product Needs

Location	Product	Amount (kg)	Need/Time	Use
Moon	O ₂	1000	Per Year	Crew Breathing - Life Support Consumable Makeup
	O ₂	3000 - 3500	2x Per Year	Non-Reusable Crew Ascent Vehicle Propulsion - Surface to Low Lunar Orbit: Earth fuel
	O ₂	~16000	2x Per Year	Reusable Ascent/Descent Propulsion - Surface to L ₁ /L ₂ : Earth Fuel (4000 kg payload)
	O_2/H_2	~30,000	2x Per Year	Reusable Ascent/Descent Propulsion - Surface to L ₁ /L ₂ (4000 kg payload)
ſ	H₂O	150,000	2x Per Year	Lunar Human Outpost & Reusable Transportation
l	O_2/H_2	150,000	Per Year	Anount needed for Propellant Delivery to LDRO for Human Mars Mission
Mars	O2/CH4 O2/CH4 H2O H2O H2O H2O	22,728/6978 59,000/17,100 3,075 15,700 38,300	Per Use/1x 480 Days Per Use/1 or2x Per Yr Surface/500 Days Per Use/1x 480 Days Per Use/1 or2x Per Yr	Non-Reusable Crew Ascent Vehicle Propulsion - Surface to High Mars Orbit Reusable Ascent/Descent Propulsion - Surface to Mars Orbit Life Support System Closure Extracted H ₂ O to Make Non-Reusable Ascent Vehicle Propellant Extracted H ₂ O to Make Reusable Ascent/Descent Vehicle Propellant



ISRU is Similar to Establishing Remote Mining Infrastructure and Operations on Earth





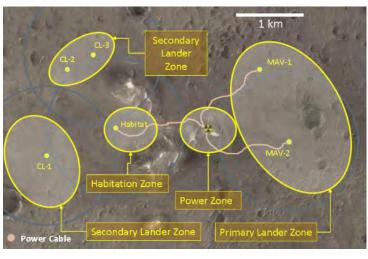


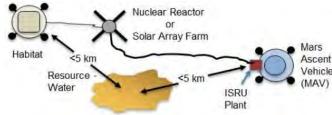
ISRU Products, Operations, and Resources Grow As Mission Needs and Infrastructure Grow



Initial Conditions:

- Hardware delivered by multiple landers before crew arrives; Multiple landing zones
- Elements offloaded, moved, deployed, and connected together remotely
- 12-18 month stay for crew of 4 to 6; Gaps of time between missions where crew is not present
- Each mission delivers extra hardware & logistics

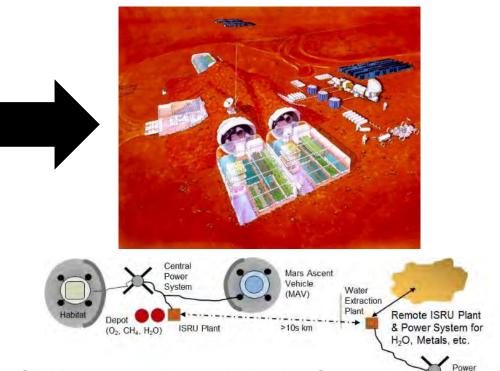




- ISRU hardware integrated with Landers
- 'Easy' Resource very close to landing site/Ascent vehicle

Ultimate Goal

- Consolidated and integrated infrastructure
- Indefinite stay with larger crews
- Roam (and mine) anywhere within 200 km diameter Exploration Zone
- Earth independence; *In situ* ability to grow infrastructure: power, habitation, food, parts, etc.



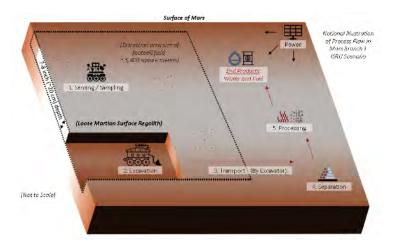
- ISRU Plants consolidated with Product Storage
- Civil Engineering and In Situ Construction operations
- Resources can now be farther from Habitat and Ascent Vehicle
- More/different resources needed for Earth independence



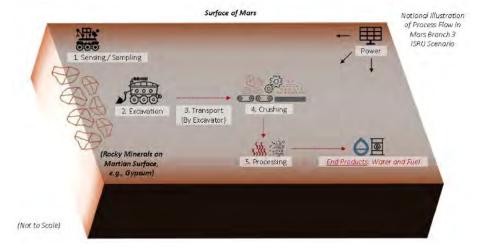
Extra-Terrestrial Mining Operations Under Consideration



Granular Soil Resource

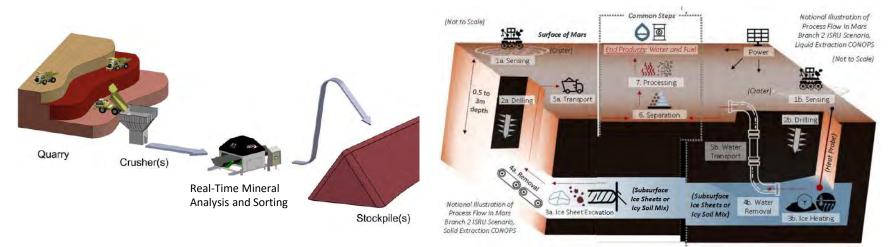


Quarry Mining



Hard Mineral Resource

Icy Resource/Subsurface Mining



Asteroid mining (not shown here) under micro-gravity conditions may require unique mining technologies and operations compared to terrestrial and Moon/Mars surface operations



Key Considerations in Pursuing Terrestrial or Space Mining



		Current Similarities/Differences
Equipment Requirements	Mass, complexity, and scale required for resource extraction, transfer, and processing	 Mass in not as important for terrestrial mining. Scale of space mining currently significantly smaller Minimizing complexity is important for both
Infrastructure Requirements	Support capabilities necessary for comm., nav., power, maintenance, personnel, and operations	 Minimizing infrastructure needs and time to establish infrastructure capabilities are critical for both Similar power, communication, and personnel needs
Energy Required 4	Type and amount of energy necessary for extraction & processing	 Energy efficiency more important for space mining Solar/renewable energy/power systems are more important for space mining
Transportation	Type, capability, frequency, and cost of transportation required to support operations and to ship products	 Minimizing transportation is important to both Shipment of cryogenic products more difficult than water or minerals
Location & Environment Adaptability	Adaptability of existing equipment and infrastructure to extreme temperatures and remote locations	 Adapting and operating in extreme temperature and abrasive environments is important to both Space mining has more extreme environments
Level of Autonomy Needed	Ability of equipment to function/operate with minimal or no oversight	 Tele-operation capabilities important to both Autonomy more important for space mining due to limited crew availability & communication time delays
Maintenance & Logistics Requirements	Level of equipment degradation/failure expected; Spares and personnel availability	 Minimizing logistics/spares is important to both for remote locations Minimizing maintenance more important for space mining due to limited crew availability
Environmental Impact & Regulations	Immediate and long-term impact on local environment; Regulations and restrictions on processing & operations	 Environmental impact, regulations, and restrictions are more important to terrestrial mining Planetary Protection rules unique to Space Mining

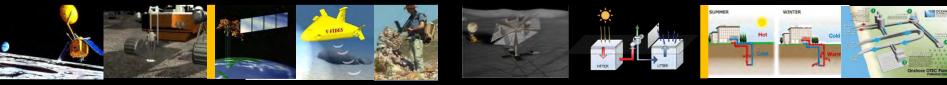


There are A lot of Similarities between ISRU and Terrestrial Applications



Prospecting for Resources





Mining for Resources



Resource Processing (Gases, Liquids, Solids)



Civil Engineering & Construction



Alternative Energy (Fuel Cells & Trash to HC)





Product Liquefaction, Storage, and Transfer



Remote Operations & Maintenance





ISRU Development and Implementation Challenges



Space Resource Challenges

- What resources exist at the site of exploration that can be used?
 - Are there enough of the right resources; Return on Investment
- What are the uncertainties associated with these resources?
 - Form, amount, distribution, impurities/contaminants
- How to address planetary protection requirements?

ISRU Operation Challenges

- How to operate in extreme environments, including temperature, pressure, dust, and radiation?
- How to achieve long duration, autonomous operation and failure recovery?
- How to operate in low gravity or micro-gravity environments?
 - Anchoring/weight-on-bit
 - Friction, cohesion, and electrostatic forces may dominate in micro-g

ISRU Technical Challenges

- Is it technically feasible to collect, extract, and process the resource?
- How to maximize performance/minimize mass
- How to achieve high reliability and minimal maintenance requirements?
- How to minimize power through thermal management integration and taking advantage of environmental conditions?

ISRU Integration Challenges

- How to optimize at the architectural level rather than the system level?
- How are other systems designed to incorporate ISRU products?
- How to manage the physical interfaces and interactions between ISRU and other systems?
- How to establish and grow production and infrastructure over time to achieve immediate and long-term Returns on Investment

Overcoming these challenges requires a multi-discipline and integrated approach



Similar Needs for Terrestrial & Space Mining



Resource Prospecting

- Physical & Mineral Characterization Instrument Types
- LIBS
- GPR
- Raman/IR
- XRD/XRF
- Hyperspectral
- Shear Vane/Cone Penetrometer
- Miniaturization and Ruggedness of Instruments
- Data Integration, Display, and Analysis of Resources

Mining

- Mine Operation Planning Tools
- Mining Technologies
- Excavation
- Drilling
- Consolidated Material Cutting/ Fracturing
- Crushing/Sorting
- Mineral BeneficiationTransport
- Environmental Compatibility
- Design for Thermal Extremes
- Material Selection
- Lubricants
- Wear Resistant Coatings
- Equipment Testing Under Realistic Conditions
- Soil Bins/Controlled Testing
- Analog Test Sites/In-Mine Testing
- Environmental Simulation Facilities
- Actual or Simulated Materials (Simulants)

Processing

- Atmosphere Collection
- Gas Compression
- Atmosphere Filtration
- Chemical Processing
- Hydrogen Production
- Syngas Production and Conversion
- CO/CO₂ Conversion to Fuel and Plastics
- Solids Processing
- Granular Material Drying
- Wear-Resistant Valves
- Metal Extraction (Oxygen Release)
- Mineral Electrolysis
- Acid Extraction
- Biological Extraction

Remote Operations

- Mining Tele-operations
- Approaches and Human Interfaces
- Same as Mining Autonomy
- Mining Autonomy
- Approaches
- Avionics, Software, Instruments, Sensors, & Cameras Needed
- Communications
 Infrastructure:
 Wireless, Bandwidth,
 Delays

Space Mining Needs

- Low Mass
- High Levels of Autonomy
- Low Maintenance/Long Life
- Modular, Multi-Mission Infrastructure
- Plug-and-Play
- High Density/Regenerable Energy All Electric
- Electro-Mechanical Actuators vs Hydraulics
- Fuel Cells/Batteries vs Combustion Engines

Product Storage and Transfer

 Liquefaction for Oxygen and Hydrogen



ISRU Has Common Challenges with Terrestrial Industry





Severe Environments

- Extreme temperatures
- Large changes in temperature
- Dust and abrasion
- No pressure vs Extreme pressure
- Environmental compatibility testing and facilities

Maintenance

- Minimal maintenance desired for long operations
- Performing maintenance is difficult in environments
- Minimize logistics inventory and supply train

Operations/Communication

- Autonomous and tele-operation;
- Delayed and potentially non-continuous communication coverage
- Local navigation and position information

Integration and Infrastructure

- Hardware from multiple countries must be compatible
- Common standards; Common interface
- Optimize at the architecture/ops level vs the individual element
- Establish and grow production and infrastructure over time to achieve immediate and long-term Returns on Investment

Return on Investment

- Need to have a return on investment to justify expense and infrastructure buildup
- Multi-use: space and terrestrial applications



Assessment of Areas of Potential Collaboration with Terrestrial Industry



3	Equipment Requirements	Mass, complexity, and scale required for resource extraction, transfer, and processing	Moderate. Collaborate on mining approaches, designs and modeling
\checkmark	Infrastructure Requirements	Support capabilities necessary for comm., nav., power, maintenance, personnel, and operations	Good. High potential for collaboration on common needs
3	Energy Required 9	Type and amount of energy necessary for extraction & processing	Moderate. Closer look required on areas of potential collaboration. Examine <i>Environmental Impact</i>
×	Transportation	Type, capability, frequency, and cost of transportation required to support operations and to ship products	
\checkmark	Location & Environment Adaptability	Adaptability of existing equipment and infrastructure to extreme temperatures and remote locations	Good. Strong need for experience, approaches and facilities for space mining
	Level of Autonomy Needed	Ability of equipment to function/operate with minimal or no oversight	Good. High potential for collaboration on common needs
3	Maintenance & Logistics Requirements	Level of equipment degradation/failure expected; Spares and personnel availability	Moderate. Closer look required on operations and failure modes. Tie to <i>Equipment Requirements</i>
×	Environmental Impact & Regulations	Immediate and long-term impact on local environment; Regulations and restrictions on processing & operations	Pa. 17





- Most Prospecting, Excavation, and Consumable Production technologies, systems, and technologies have been shown to be feasible at subscale and for limited test durations
- Drivers
 - Hardware simplicity and robustness are as important as minimizing mass and power
 - Hardware commonality with other systems (propulsion, power, life support, thermal) can significantly reduce costs and logistics
- Work still required to:
 - Acquire, fuse, and analyze site, environment, and resource information for site selection and resources assessment
 - Perform prospecting missions to better define resources
 - Scale up technologies and operations to meet production and processing rates for human mission needs (*pilot scale for terrestrial industry*)
 - Operate hardware and systems under relevant mission environments (environmental chambers and analog sites); Understand how to take advantage of the environment and day/night cycle
 - Perform long-duration testing to understand hardware life, maintenance, and logistics needs
 - Add autonomy to operations, especially for mining operations

Partnering with Terrestrial Industry and co-leveraging hardware is important to NASA





- Maintain and expand dialog with Industry
 - Examine similarities in Key Considerations and Similar Needs
 - Address Common Challenges
 - Examine differences in Key Considerations to understand potential paradigm changes/technology infusion
- Examine use of Test Facilities and Approaches; especially for Environmental Compatibility
- Target 'Spin-in/Spin-off' Technology Relationships in Areas of Potential Collaboration
 - Procurements/Request for Proposals (RFPs)
 - Cooperative Agreements
 - Space Act Agreements





BACKUP





• 'Resources'

- Traditional: Water, atmospheric gases, volatiles, solar wind volatiles, metals, alloys, etc.
- Non-traditional: Trash and wastes from crew, spent landers and residuals, etc.

Energy

- Thermal Energy Storage Using Modified Regolith
 - Thermal conductivity of unmodified lunar regolith is very low (~1 mW/m-K); good insulator.
- Permanent/Near-Permanent Sunlight
 - Stable thermal control & power/energy generation and storage
- Permanent/Near-Permanent Darkness
 - Thermal cold sink for cryo fluid storage & scientific instruments

Environment

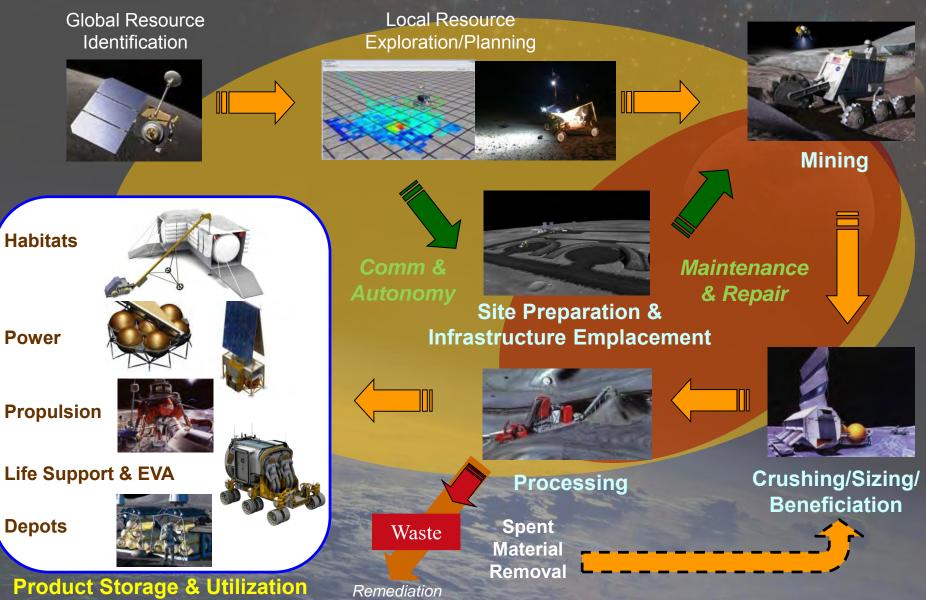
- Vacuum
- Micro/Reduced Gravity
- Large Thermal Gradients
- Atmosphere Drag

Location

- Stable Locations/'Real Estate':
 - Earth viewing, sun viewing, space viewing, staging locations
- Isolation from Earth
 - Electromagnetic noise, hazardous testing & development activities (nuclear, biological, etc.), extraterrestrial sample curation & analysis, storage of vital information, etc.

Space 'Mining' Cycle: *Prospect to Product*

Resource Assessment (Prospecting)

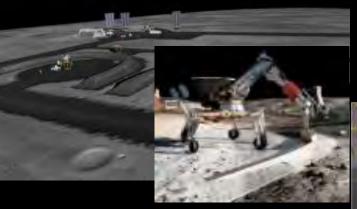


Lunar ISRU Mission Capability Concepts

Excavation & Regolith Processing for O₂ Production

Resource Prospecting – Looking for Polar Ice

Carbothermal Processing with Lunar Lander Assets



Landing Pads, Berms, Roads, and Structure Construction

Thermal Energy Storage Construction





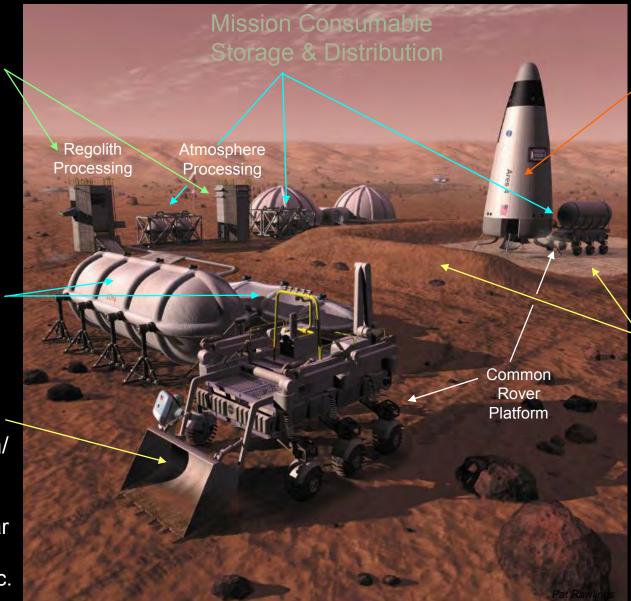
Consumable Depots for Crew & Power

Mars ISRU Mission Capability Concepts

Resource Processing Plants

Collapsible/ Inflatable Cryogenic Tanks

Multi-use Construction/ Excavator: resources, berms, nuclear power plant placement, etc.



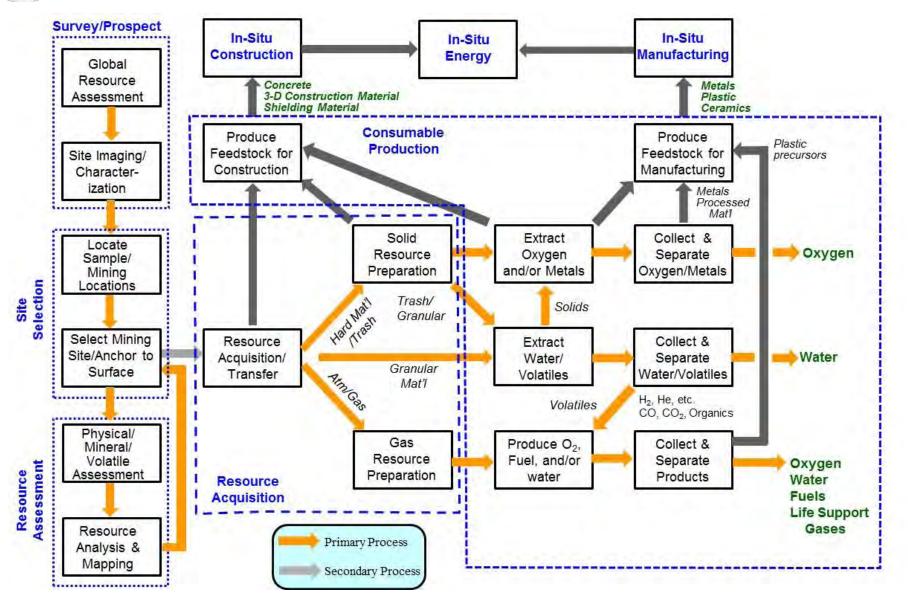
Reusable lander/ascent vehicle or surface hopper fueled with in-situ propellants

Landing pad & plume exhaust berm



ISRU Capability-Function Flow Chart

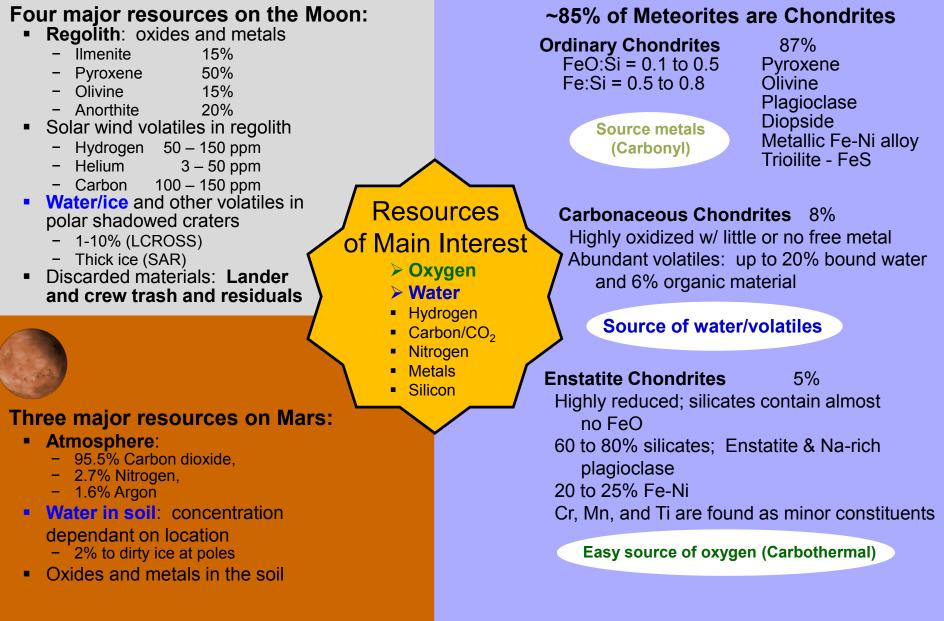






Natural Space Resources







Lunar Resource Overview



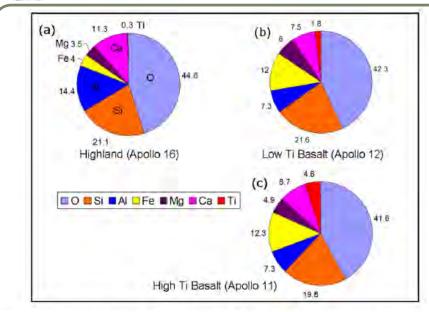


Figure 3. Example chemical compositions of lunar soils: (a) lunar highland minerals (Apollo 16); (b) low-Ti basalts (Apollo 12); and (c) high-Ti basalts (Apollo 11). Based on data collated by Stoeser et al. (2010), and reprinted from *Planetary and Space Science*, Vol. 74, Schwandt C, Hamilton JA, Fray DJ and Crawford IA, 'The production of oxygen and metal from lunar regolith' 49-56, Copyright (2012), with permission from Elsevier.

Table 1. Average concentrations of solar wind implanted volatiles in the lunar regolith (Fegley and Swindle 1993), where the quoted errors reflect the range (\pm one standard deviation) of values found at different sampling locations. The corresponding average masses contained within 1 m³ of regolith (assuming a bulk density of 1660 kg m⁻³; Carrier et al., 1991) are also given.

Volatile	Concentration ppm (µg/g)	Average mass per m ³ of regolith (g)
H	46 ± 16	76
³ He	0.0042 ± 0.0034	0.007
⁴ He	14.0 ± 11.3	23
С	124 ± 45	206
N	81 ± 37	135
F	70 ± 47	116
C1	30 ± 20	50

In addition to the volatiles listed in Table 1, lunar soils contain small quantities (typically $\leq 1 \ \mu g/g$) of the solar wind derived noble gasses Ne and Ar (and much smaller quantities of Kr and Xe). Perhaps more interesting from a resource perspective, they also contain a significant quantity of sulphur (715±216 $\mu g/g$; Fegley and Swindle 1993), mostly derived from the mineral troilite (FeS), and this would probably also be released by any process which extracts the other volatile elements.

From "Lunar Resources: A Review" by Ian Crawford, 2015

		Lunar Basalt	Lunar Breccias	Lunar Soil	Earth Crust	-			Lunar Basalt	Lunar Breccias	Lunar Soil	Earth Crust
Pr	ppm	13		7	9.2		Ag	ppb	1.5	18	9	75
Nd	ppm	63	40	39	41.5		Cd	ppb	10	100	50	150
Sm	ppm	21	14	13	7.05		In	ppb	3	5	<10	25
Eu	ppm	2.2	1.9	1.7	2		Те	ppb	16	72		1
Gd	ppm	27	20	15	6.2		Se	ppm	0.7	1.6	0.8	0.05

Rare Earth Elements

From Bob Wegeng/PNNL

Vapor Mobilized Elements



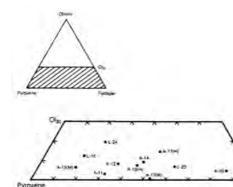


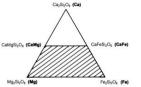
TABLE 5.1. Modal proportions (vol.%) of minerals and glasses in soils from the Apollo (A) and Luna (L) sampling sites (90-20 µm fraction, not including fused-soil and rock fragments).

				A-	A-		A-	A-			
	A-	A-	A-14	(H)	(M)	A-16	(H)	(M)	L-16	L-20	L-
Plagioclase	21.4	23.2	31.8	34.1	12.9	69.1	39.3	34.1	14.2	52.1	20.9
Pyroxene	44.9	38.2	31.9	38.0	61.1	8.5	27.7	30.1	57.3	27.0	51.6
Olivine	2.1	5.4	6.7	5.9	5.3	3.9	11.6	0.2	10.0	6.6	17.5
Silica	0.7	1.1	0.7	0.9	-	0.0	0.1	-	0.0	0.5	1.7
Ilmenite	6.5	2.7	1.3	0.4	0.8	0.4	3.7	12.8	1.8	0.0	1.0
Mare Glass	16.0	15.1	2.6	15.9	6.7	0.9	9.0	17.2	5.5	0.9	3.4
Highland Glass	8.3	14.2	25.0	4.8	10.9	17.1	8.5	4.7	11.2	12.8	3.8
Others	-	-	-	-	2.3	-	-	0.7	-	-	
Total	99.9	99.9	100.0	100.0	100.0	99.9	99.9	99.8	100.0	99.9	99.9

Data from *Papike et al.* (1982), *Simon et al.* (1982), *Laul et al.* (1978a), and Papike and Simon (unpublished). (H) Denotes highland. (M) Denotes mare.

Feldsp





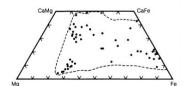


TABLE 5.2. Summary of modal data (vol.%) for mare basalts (after *BVSP*, 1981, p. 255).

	Oxide	Pyroxen	Feldspa	
	Minerals	e	r	Olivine
A-17 high Ti	24.4	47.7	23.4	4.6
A-11 high K	20.6	57.5	21.7	0.1
A-17 low K	15.1	51.6	33.3	-
A-11 low K	14.6	50.9	32.2	2.3
A-12 ilmenite	9.3	61.1	25.9	3.6
A-12 pigeonite	9.1	68.4	21.1	1.4
A-12 olivine	7.1	53.5	19.2	20.2
L-16 aluminous	7.1	51.5	41.2	0.1
A-15 olivine	5.5	63.3	24.1	7.0
A-15 pigeonite	3.7	62.5	33.8	-
A-14 aluminous	3.2	53.8	43.0	-
L-24 ferrobasalt	1.8	48.6	39.1	10.4
L-24	1.4	60.2	34.2	4.2
ferrobasalt A-17 VLT	1.0	61.7	31.9	5.4

Modal data normalized to 100% for the four phases considered, in Apollo (A) and Luna (L) samples. Ordered from top to bottom in terms of decreasing modal content of opaque oxide minerals.

contains some Mg substituting for Fe (Table A5.11), which arises from the solid solution that exists between ilmenite (FeTiO₃) and MgTiO₃, the mineral *geikielite*. Other elements are present only in minor to trace amounts (i.e., <1%); these include Cr, Mn, Al, and V. In addition, ZrO₂ contents of up to 0.6% have





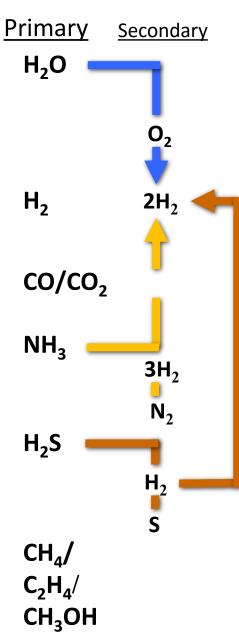
						Instru	ment	
	Column Density (# m ⁻²)	Relative to H2O(g) (NIR spec only)	Concentration (%)	Long-term Vacuum Stability Temp (K)	UV/Vis	NIR	LAMP	М3
CO	1.7e13±1.5e11		5.7	15			x	
H ₂ O(g)	5.1(1.4)E19	1	5.50	106		x		
H ₂	5.8e13±1.0e11		1.39	10			x	
H ₂ S	8.5(0.9)E18	0.1675	0.92	47	x	x		
Са	3.3e12±1.3e10		0.79				x	
Hg	5.0e11±2.9e8		0.48	135			x	
NH ₃	3.1(1.5)E18	0.0603	0.33	63		x		
Mg	1.3e12±5.3e9		0.19				x	
SO ₂	1.6(0.4)E18	0.0319	0.18	58		x		
C ₂ H ₄	1.6(1.7)E18	0.0312	0.17	~50		x		
CO ₂	1.1(1.0)E18	0.0217	0.12	50	x	x		
CH₃OH	7.8(42)E17	0.0155	0.09	86		x		
CH ₄	3.3(3.0)E17	0.0065	0.04	19		x		
ОН	1.7(0.4)E16	0.0003	0.002	>300 K if adsorbed	x	x		x
H ₂ O (adsorb)			0.001-0.002					x
Na		1-2 kg		197	x			
CS					x			
CN					x			
NHCN					x			
NH					x			
NH ₂					x			

Volatiles comprise possibly 15% (or more) of LCROSS impact site regolith

*Chart courtesy of Tony Colaprete







- Drinking water, washing, EVA coolant, radiation shielding
- Steam reforming of hydrocarbons
- Life support/breathing
- Rocket oxidizer, fuel cell reactant
- Rocket fuel, fuel cell reactant
- Key reactant in petro-chemical industry
- Key reactant in hydrocarbon rocket fuel & petro-chemical industry
- Thermal control fluid, cleaning agent
- Fuel cell reactant
- Life support buffer gas, inert gas for science, purge gas
- Contaminant; potential production of sulfuric acid
- ????
- Rocket fuel, fuel cell reactant
- Key reactant in hydrocarbon rocket fuel & petro-chemical industry



Mars Resources



Resource	Potential Minera	al Source	Reference				
Water, Hydration/ Hydroxyl	Gypsum – (CaSO ₄ .2H ₂ Jarosite – (KFe ³⁺ ₃ (OH) Opal & hydrated silica Phyllosilicates Other hydrated miner	6(SO4) ₂) a – (SiO ₂ .nH ₂ O)	Horgan, et al.(2009), Distribution of hydrated minerals in the north polar region of Mars, J. Geophys. Res., 114, E01005 Mustard et al.(2008), Hydrated silicate minerals on Mars observed by the Mars Reconnaissance Orbiter CRISM instrument, Nature 454, 305-309				
Water, Ice	Icy soils Glacial deposits		Mellon & Feldman (2006) Dickson et al. (2012)				
Iron*	Hematite Magnetite Laterites	Jarosite Triolite Ilmenite	Ming et al. (2006), Geochemical and mineralogical indicators for Aqueous process in Columbia Hills of Gusev Crater, Mars ²⁰ JGR 111, E02S12 Poulet et al. (2007), Martian surface mineralogy from OMEGA/Mex: Global mineral maps ²⁰ JGR 112, E08S02				
Aluminum*	Laterites Aluminosilicates	Plagioclase Scapolite					
Magnesium*	Mg-sulfates, Mg-rich	olivines, Forsterite					
Silicon	Pure amorphous silica Hydrated silica Phyllosilicates	a	Rice et al. (2010), "Silica-rich deposits and hydrated minerals at Gusev Crater, Mars: Vis-NIR spectral characterization and regional mapping" lcarus 205 (2010) 375–395				
Titanium*	Ilmenite, Titanomagn	etite	Ming et al. (2006), JGR 111, E02512				

	Oxides (Wt%)											Elements (ppm)					
	SiO ₂	TiO ₂	Al ₂ O ₃	FeO	MnO	MgO	CaO	Na ₂ O	K ₂ O	P ₂ O ₅	Cr ₂ O ₃	Cl	SO ₃	Ni	Zn	Br	Ge
MER Spirit – Laguna Soils, Panda Subclass	46.8	0.79	10.5	16.1	0.33	9.6	6.2	3	0.38	0.75	0.35	0.6	4.6	684	190	42	6
Rocknest Soil (Portage)	43.0	1.2	9.4	19.2	0.42	8.7	7.3	2.7	0.49	0.95	0.49	0.69	5.5	456	326	34	
Mojave Mars Simulant	t 49.4	1.09	17.1		0.17	6.1	10.5	3.3	0.48	0.17	0.05		0.1	118	71		0.07

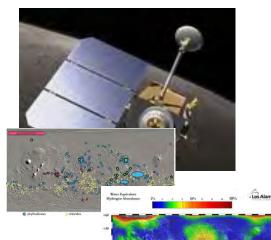


Approach to Resource Assessment



Remote Assessment

Orbiters



Goals:

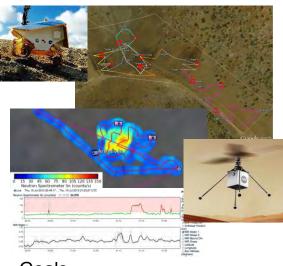
- Obtain data on terrain, minerals, and water resources to select landing sites of consideration
- ii. Obtain data at resolution to plan surface Exploratory Assessment of terrain and resources

Instruments

- Better mineral resolution for chemistry and hydration
- Passive and active subsurface hydrogen and layer

Exploratory Assessment

Rovers, Hoppers, Aerial Vehicles, Impactors, Instrumented Landers



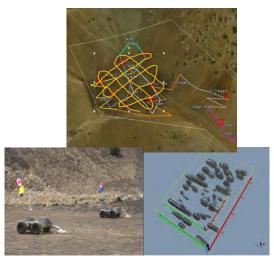
Goals:

- Obtain data on physical/mineral characteristics and water/volatiles.
- Obtain sufficient data to determine if the site warrants a Focused Assessment of resources

Instruments

- Should cover physical/geotech, chemical/mineral, and volatile characterization
- Passive and active subsurface assess

Focused Assessment, Mapping, & Planning Rover or Crew



Goals:

- i. Ensure sufficient resources exist in form and location expected
- ii. Build 3-D interpretation of data to define resource for mining operations

Instruments

- Should cover physical, chemical/mineral, and volatile characterization
- Passive and active subsurface assess

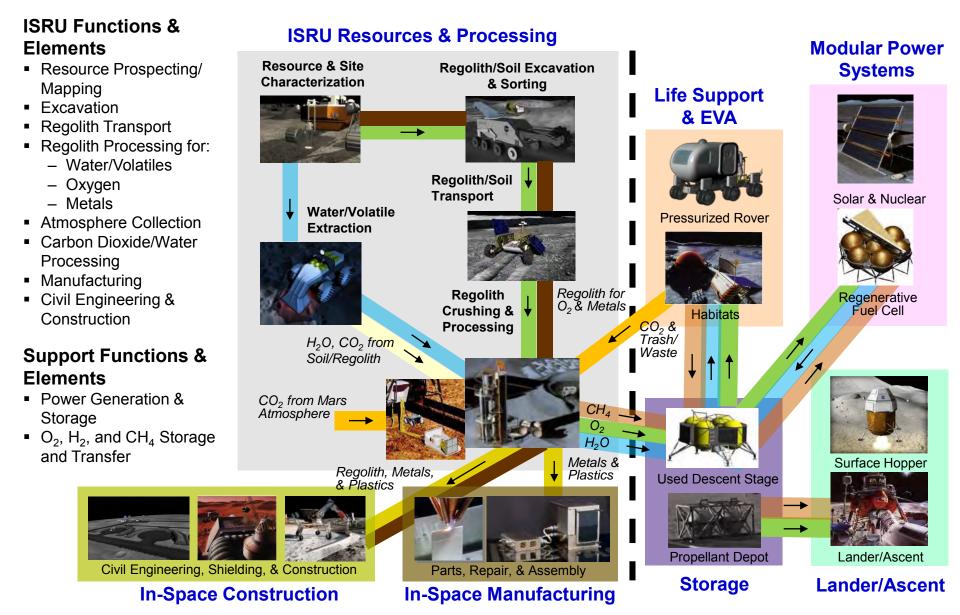
The extent of Resource Assessment performed is a function of Knowledge Needed, Risk Acceptability, Funding Availability, and Time



ISRU Integrated with Exploration Elements

(Mission Consumables)



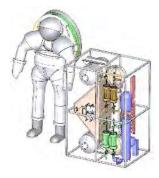




Mars Atmosphere & Water Resource Attributes



Atmosphere Processing



Atmosphere

- Pressure: 6 to 10 torr (~0.08 to 0.1 psi);
- >95% Carbon Dioxide
- Atm. temperature: +35 C to -125 C
- Everywhere on Mars;

Lower altitude the better

 Chemical processing similar to life support and regenerative power

Granular Regolith Processing for Water



Mars Garden Variety Soil

- Low water concentration 1-3%
- At surface
- Granular; Easy to excavate
- 300 to 400 C heating for water removal
- Excavate and transfer to centralized soil processing plant
- Most places on Mars; 0 to +50 Deg. latitude

Gypsum/Sulfate Processing for Water



Gypsum or Sulfates

- Hydrated minerals 5-10%
- At Surface
- Harder material: rock excavation and crushing may be required
- 150 to 250 C heating for water removal
- Localized concentration in equatorial and mid latitudes

Subsurface Ice

90%+ concentration

Icy Regolith

Processing for Water

- Subsurface glacier or crater: 1 to 3 m from surface possible
- Hard material
- 100 to 150 C heating for water removal
- Downhole or on-rover processing for water removal
- Highly selective landing site for near surface ice or exposed crater; >40 to +55 Deg. latitude

Increasing Complexity, Difficulty, and Site Specificity



ISRU Product/Resource Processing Options Under Consideration



Oxygen/Fuel Production from Mars Atmosphere

Atmosphere Collection

- Dust Filtration
- Gas Separation (CO₂, N₂, Ar)
- Gas Pressurization (0.1 to >15 psia)
 - Pumps/Compressors
 - Cryogenic Separation
 - Adsorption

Chemical Processing

- CO₂ Reduction
 - Solid Oxide Electrolysis
 - Reverse Water Gas Shift
 - Ionic Liquid/PEM
 Electrolysis
- Fuel Production
 - Sabatier (CH₄)
 - Fischer Tropsch
 - Alcohols
 - Ethylene \rightarrow Plastics
- Water Processing
 - Water Electrolysis (PEM vs SOE)
 - Water
 Cleanup/Deionization

Water/Volatile Extraction From Soils

Solid Extraction and Transport

- Granular Soil Excavation/Extraction
 - Drills/Augers (1 to 3 m)
 - Load/Haul/Dump (LHD)
 - Bucket Wheels/Drums
- Consolidated Material Extraction & Preparation
 - Drills/Augers
 - Percussive Blades
 - Ripper & LHD
 - Crushing & Sorting
- Material Transfer
 - Augers
 - Pneumatic
 - Bucket ladders

Water/Volatile Extraction

- Hydrated soils
 - Open Reactor/Heating
 - Closed Fluidized Reactor
 - Auger Dryer
- Icy soils
 - Transport to Reactor
 - Downhole Enclosure
 - Downhole Heating & Removal

Oxygen Extraction from Minerals

- Hydrogen Reduction of Iron Oxides
- Methane Reduction of Silicates
- Molten Oxide Reduction

Metal Extraction from Minerals

- Molten Oxide Reduction
- Molten Salt Reduction
- Ionic Liquids/Acids
- Biological Extraction

Oxygen Extraction from Minerals

Metal Extraction from Minerals

NASA

Current NASA ISRU Missions Under Development



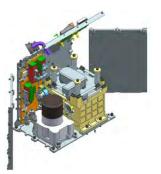


Resource Prospector – RESOLVE Payload

- Measure water (H₂O): Neutron spec, IR spec., GC/MS
- Measure volatiles H₂, CO, CO₂, NH₃, CH₄, H₂S: GC/MS
- Possible mission in 2020

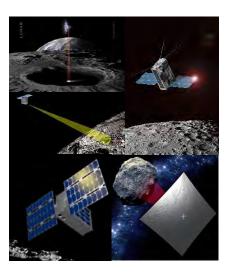
Cubesats (SLS EM-1 2018)

- Lunar Flashlight: Uses a Near IR laser and spectrometer to look into shadowed craters for volatiles
 Lunar IceCube: Carries the Broadband InfraRed Compact High Resolution Explorer Spectrometer (BIRCHES)
 LunaH-MAP: Carries two neutron spectrometers to produce maps of near-surface hydrogen (H)
 Skyfire: Uses spectroscopy and thermography for surface characterization
- NEA Scout: Uses a science-grade multispectral camera to learn about NEA rotation, regional morphology, regolith properties, spectral class



Mars 2020 ISRU Demo

- Make O₂ from Atm. CO₂: ~0.01 kg/hr O₂; 600 to 1000 W-hrs; 15 sols of operation
- Scroll Compressor and Solid Oxide Electrolysis technologies
- Payload on Mars 2020 rover







Every 1 kg of propellant made on the Moon or Mars saves 7.4 to 11.3 kg in LEO

Mars mission

Potential 334.5 mT launch mass saved in LEO = 3 to 5 SLS launches avoided per Mars Ascent

Oxygen only
Methane + Oxygen

Phobos mission

– Trash to O_2/CH_4

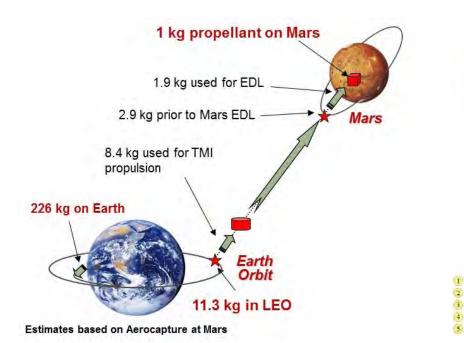
75% of ascent propellant mass; 20 to 23 mT 100% of ascent propellant mass: 25.7 to 29.6 mT Regeneration of rover fuel cell reactant mass

Lunar Destination Orbit

Lunar Surface Lunar Rendezvous Orbit

Earth Surface

1000+ kg of propellant



A Kilogram of Mass Delivered Here	Adds This Much Initial Architecture Mass in LEO	Adds This Much To the Launch Pad Mass
Ground to LEO	-	20.4 kg
LEO to Lunar Orbit (#1→#2)	4.3 kg	87.7 kg
LEO to Lunar Surface (#1→#3; e.g., Descent Stage)	7.5 kg	153 kg
LEO to Lunar Orbit to Earth Surface (#1#4#5, e.g., Orion Crew Module)	9.0 kg	183.6 kg
Lunar Surface to Earth Surface (#3	12.0 kg	244.8 kg
LEO to Lunar Surface to Lunar Orbit (#1-#3#4; e.g., Ascent Stage)	14.7 kg	300 kg
LEO to Lunar Surface to Earth Surface (#1→#3→#5; e.g., Crew)	19.4 kg	395.8 kg